

# ***GAS COOLING OF DIVERTORS***

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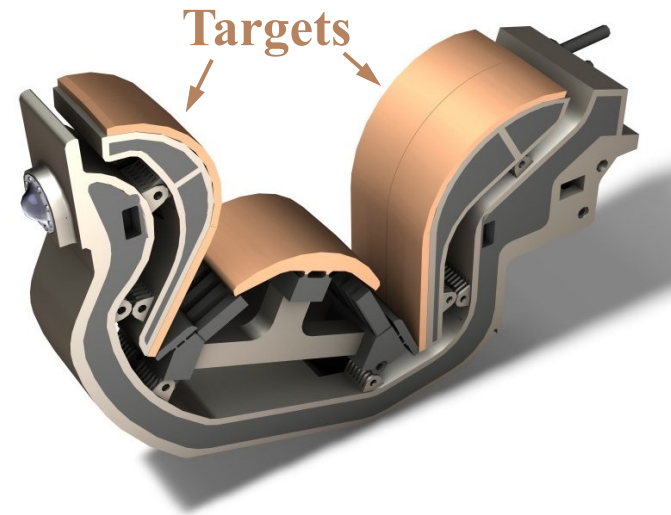
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# INTRODUCTION

- **Power handling/exhaust in plasma-material interactions (PMI)**
  - Divertor configuration: confine core plasma with poloidal magnetic field
    - Exhaust ~20% of energy onto divertor targets (plasma-facing components, or PFC) near X-point(s)
    - Targets remove plasma impurities and fusion products from scrape-off layer, converting kinetic energy to heat  $\Rightarrow$  subject to extremely high heat fluxes
  - ITER divertor specifications
    - Steady-state heat fluxes  $q'' = 10 \text{ MW/m}^2$
    - “Slow transients”:  $q'' = 20 \text{ MW/m}^2$  over 10 s
  - Lower ( $< 2 \text{ MW/m}^2$ ) heat load on first wall
  - Actual divertor (and first wall) heat load specifications TBD
  - Candidate coolants for long-pulse reactors: He, water (DEMO), liquid metals



# GEORGIA TECH STUDIES

**Aim: evaluate thermo-fluids performance of gas-cooled solid-tungsten (W) divertors**

■ **Experiments complemented by numerical simulations**

- Tests in He loop on stainless + WL10 test sections: consider pressure boundary (vs. W PFC/tile) at prototypical pressures, nearly prototypical temperatures and heat fluxes
  - Measure cooled surface temperatures  $T_s$ , pressure drops  $\Delta p$ , and incident heat flux  $q''$  (from He energy balance) over a range of He mass flowrates  $\dot{m}$
  - Dimensionless heat transfer coefficient, Nusselt number  $Nu$ , from  $(T_s, q'')$ : correlations for  $Nu(Re)$  [Reynolds number  $Re$  dimensionless  $\dot{m}$ ]
  - Dimensionless pressure loss coefficient  $K_L$  from  $\Delta p$ : correlations for  $K_L(Re)$
- Experimental data validate numerical models in commercial CFD software (ANSYS Workbench)
- Develop performance curves for maximum heat flux on target plates

# OVERVIEW

## Updates on current research

- **“Flat design” finger-type divertor update**
  - Revised pressure loss coefficients  $K_L$
- **CO<sub>2</sub>, vs. He, cooled divertors**
  - Thermo-fluids performance of finger-type (= “HEMJ minus the HE”) divertors
- **Helium-cooled flat plate (HCFP) experiments**
  - “Short” HCFP test section in small ( $\dot{m} \leq 10$  g/s) GT He loop

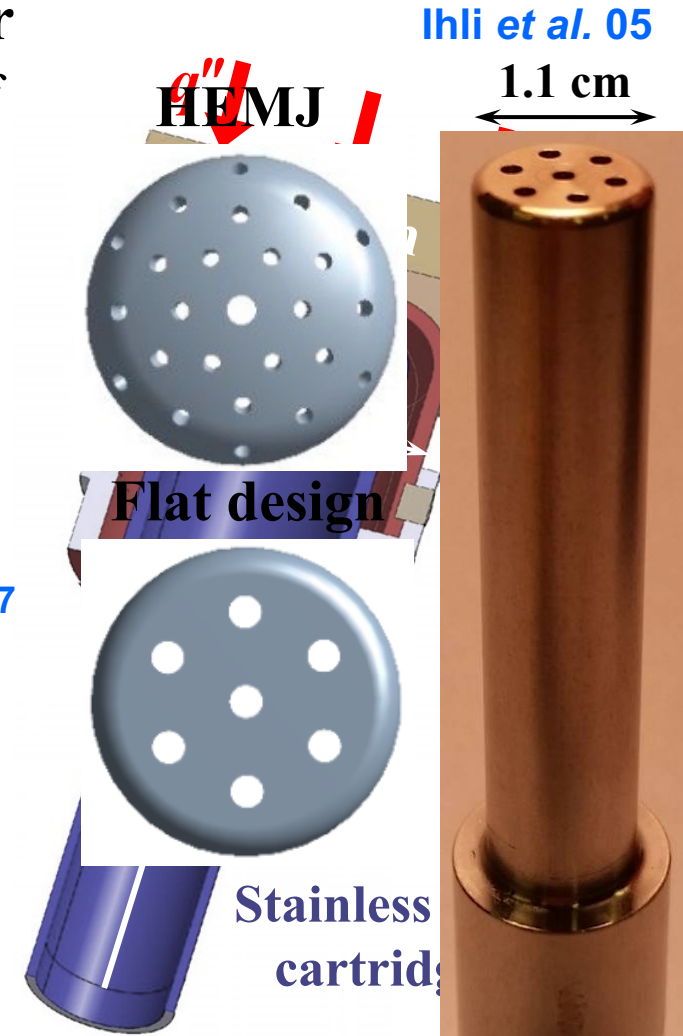
# FINGER-TYPE DIVERTORS

## ■ Helium-cooled multijet (HEMJ) divertor

- 25 (24  $\varnothing 0.6$  mm + 1  $\varnothing 1.04$  mm) impinging jets of (600 °C, 10 MPa) He cool curved inner surface of PFC across  $H = 0.9$  mm gap
- $\dot{m} = 6.8$  g/s; tile area  $\approx 2.8$  cm<sup>2</sup>

## ■ Task 1 of PHENIX: optimize cooling

- Developed “flat” design to simplify HEMJ:  
7 ( $\varnothing 1.18$  mm) impinging jets cool flat inner surface across  $H = 1.25$  mm gap
- Experiments in small GT He loop
  - Incident heat fluxes  $q'' \leq 7.1$  MW/m<sup>2</sup>
  - He conditions at inlet:  $T_i \leq 400$  °C,  $p_i \approx 10$  MPa
  - Reynolds number  $Re \leq 5.4 \times 10^4$  (vs. prototypical  $Re_p = 2.2 \times 10^4$ )



# FLAT DESIGN PARAMETERS

## ■ Reynolds number $Re$ dimensionless He mass flow rate

- Jet diameter  $D_j = 1.18$  mm; jets area  $A_j = 766.4$  mm<sup>2</sup>
- Prototypical  $Re_p = 2.2 \times 10^4$

$$Re = \frac{4}{\pi} \frac{\dot{m} D_j}{\mu(T_i) A_j}$$

## ■ Average heat transfer coefficient (HTC) $h$

- Heat flux  $q''$  from He energy balance
- $\bar{c}_p$  = specific heat evaluated at  $(T_o + T_i) / 2$
- Area of heated, cooled surfaces  $A_h = 227$  mm<sup>2</sup>,  
 $A_c = 154$  mm<sup>2</sup>
- $\bar{T}_c$  average cooled surface temperature

$$q'' = \dot{m} \bar{c}_p (T_o - T_i) / A_h$$

$$h = \frac{q'' A_h}{(\bar{T}_c - T_i) A_c}$$

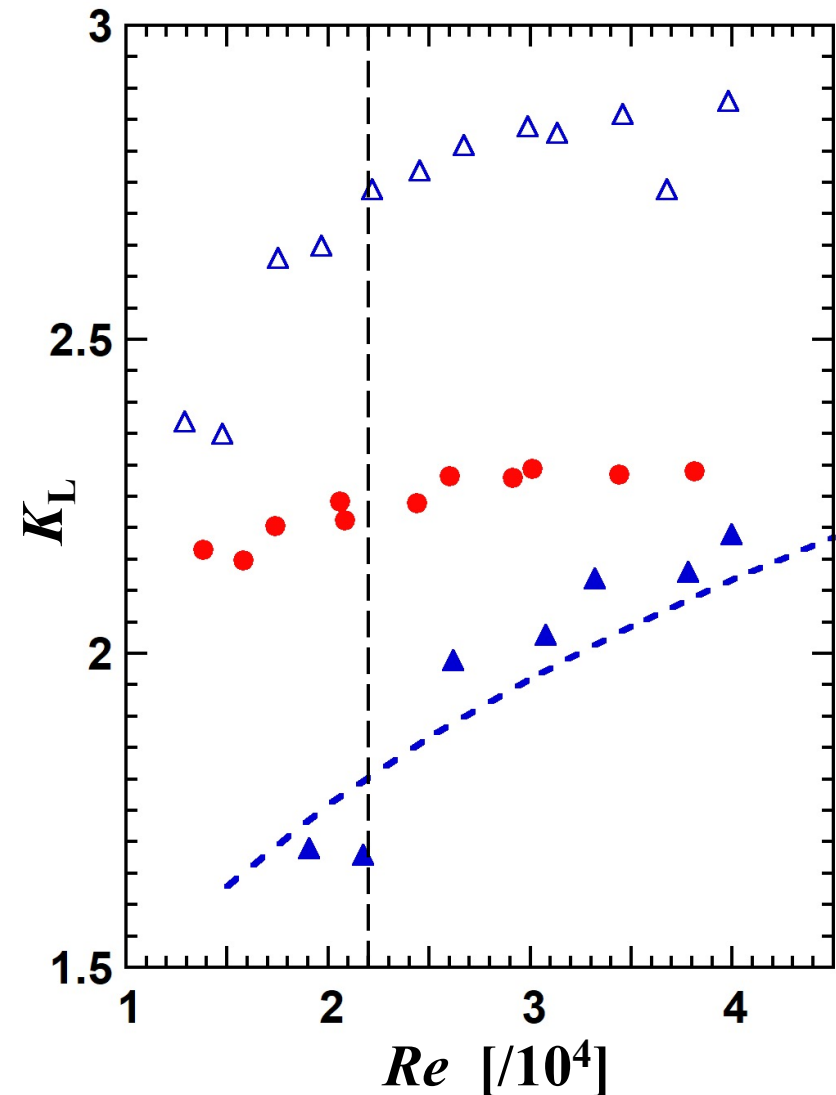
## ■ Nusselt number $Nu$ dimensionless HTC

- $\bar{k}$  = He thermal conductivity evaluated at  $(T_o + T_i) / 2$
- Develop correlation assuming power law for  $Nu(Re)$  since test section uses actual divertor materials

$$Nu = \frac{h D_j}{\bar{k}}$$

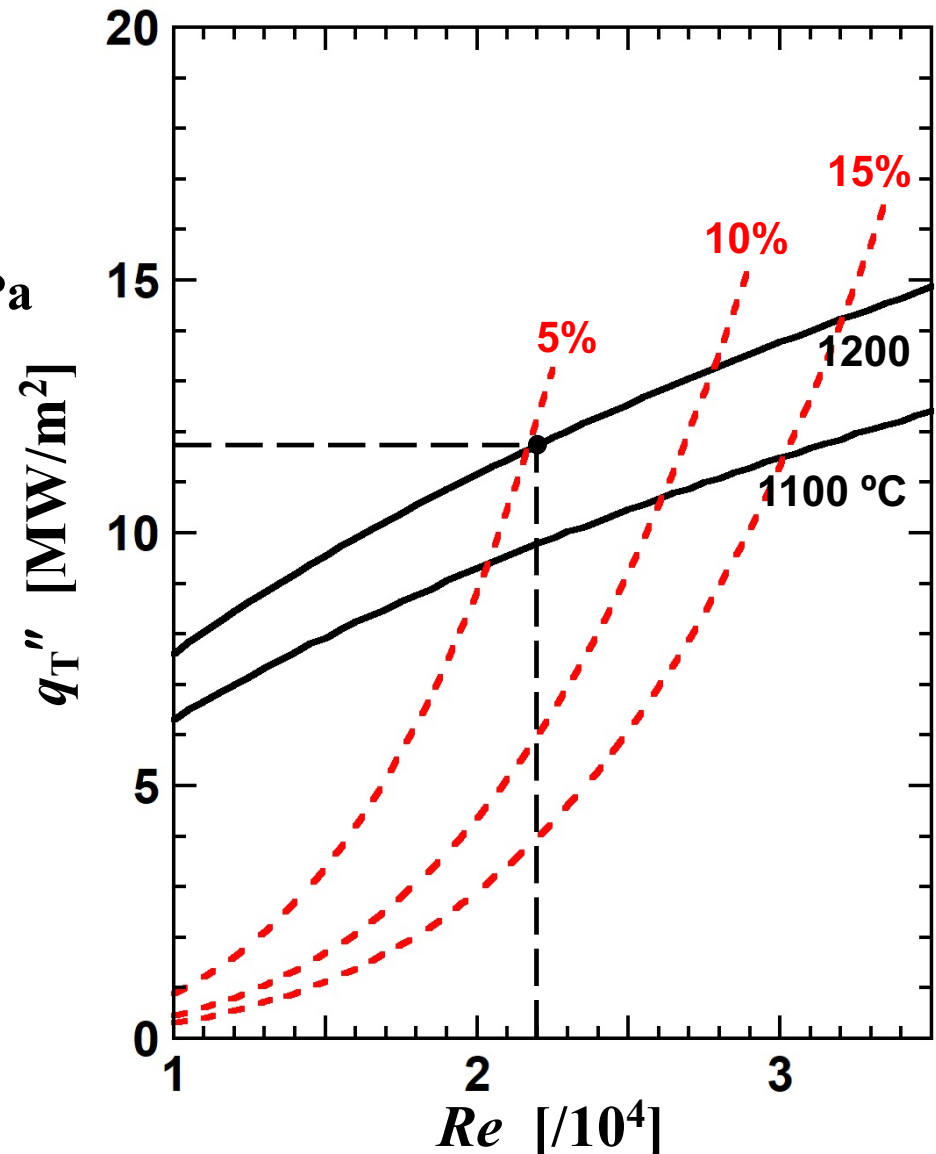
# UPDATED PRESSURE DROPS

- **Pressure loss coefficient  $K_L$**   
**dimensionless  $\Delta p$** 
  - $\bar{V}$  average speed
- **Flat design results**
  - Original  $K_L$  (●) Zhao et al. 18
  - Initial tests with new test section gave much higher  $K_L$  (△) Lee et al. 21
  - Tests with new manifold (and same test section) give lower  $K_L$  (▲): at  $Re_p$ ,  $K_L = 1.7$ , vs. 2.3 for HEMJ
  - Correlation:  $K_L = 0.125 Re^{0.267}$  (---)
- **Extrapolate  $Nu$ ,  $K_L$  correlations to prototypical conditions**
  - $Nu = 0.145 Re^{0.685}$   $T_i \geq 300$  °C



# PERFORMANCE CURVES

- Estimate maximum incident heat flux on tile  $q_T''$  ( $A_T = 281 \text{ mm}^2$ ) removed at prototypical conditions:  $T_i = 600 \text{ }^\circ\text{C}$ ,  $p \approx 10 \text{ MPa}$ 
  - Average pressure boundary temperature (based on 1D conduction)  
 $T_s = 1100 \text{ }^\circ\text{C}$ ,  $1200 \text{ }^\circ\text{C}$
  - Helium pumping power (as fraction of thermal power)  $\beta = 5\%$ ,  $10\%$ ,  $15\%$
  - At  $Re_p = 2.2 \times 10^4$ :  $q_T'' = 11.7 \text{ MW/m}^2$ ,  $\beta \approx 5\%$
- Based on updated  $K_L$  results, flat design and HEMJ have similar thermal-fluids performance





# OTHER GASEOUS COOLANTS

**Objective: evaluate supercritical carbon dioxide (sCO<sub>2</sub>) vs. He as a divertor coolant**

- **Fission reactors have used two gaseous coolants: He and CO<sub>2</sub>**
  - CO<sub>2</sub> coolant for Magnox (1950s), Gen II AGRs (1970s)
    - AGRs: dissociation of CO<sub>2</sub> into CO and O<sub>2</sub> (especially at higher temperatures > 600 °C) results in oxidation of graphite, metals
- **Focus on supercritical CO<sub>2</sub>: critical point (7.38 MPa, 31 °C)**
  - High densities ⇒ compact, high efficiency power cycles
    - Evaluation of various sCO<sub>2</sub> power cycles using heat from DEMO [Stepanek et al. 20](#)
  - Evaluate thermo-fluids performance of sCO<sub>2</sub> as (last?) part of the FESS Next Study
    - $Nu, K_L \Rightarrow$  estimate maximum heat flux and pumping power under prototypical conditions
    - Exploit existing validated numerical models of HEMJ

# sCO<sub>2</sub> SIMULATIONS

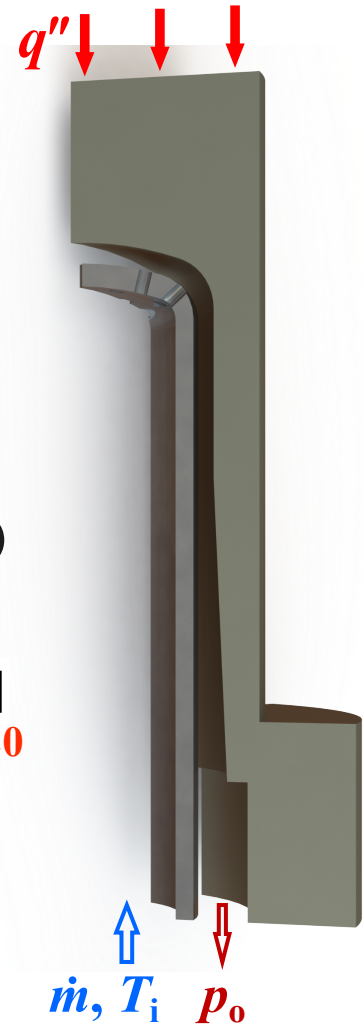
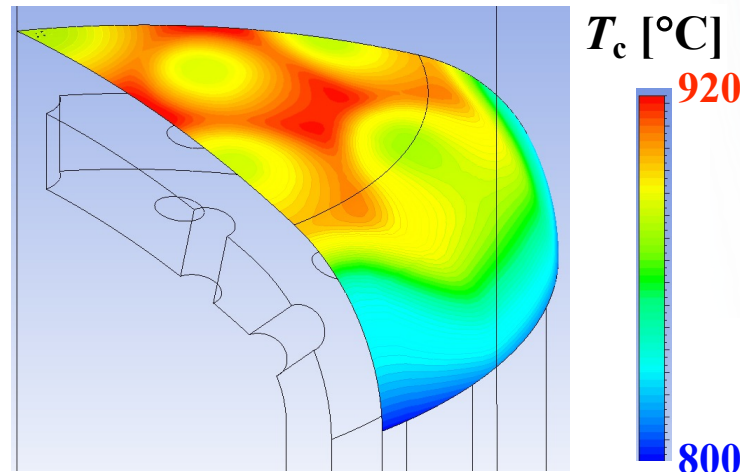
## ■ Numerical model of 60° wedge of HEMJ

- Commercial CFD software (FLUENT)
  - Unstructured hexahedral mesh with enhanced wall functions: 6.2M elements
  - Standard  $k$ - $\epsilon$  turbulence model
  - Specify  $q''$ ,  $\dot{m}$ ,  $T_i$ ,  $p_o$ ; adiabatic walls (except top wall)

## ■ Constrained by temperature limits on materials

- For sCO<sub>2</sub>, much higher mass flow rate  $\dot{m}_C = 59$  g/s ( $Re = 2 \times 10^5$ ) to match cooled surface temperatures  $T_c$  to those for He at  $Re_p = 2.2 \times 10^4$ 
  - Higher HTC with He

sCO<sub>2</sub>:  $T_i = 600$  °C,  $p_o = 10$  MPa



# sCO<sub>2</sub> PERFORMANCE CURVES

## ■ Simulation results at $p_o = 10$ MPa

- $Nu = 0.00626 Re^{0.837} \kappa^{0.19}$  ( $\kappa \equiv k_s / k$  thermal conductivity ratio), *vs.*  $Nu = 0.0377 Re^{0.687} \kappa^{0.19}$  for He

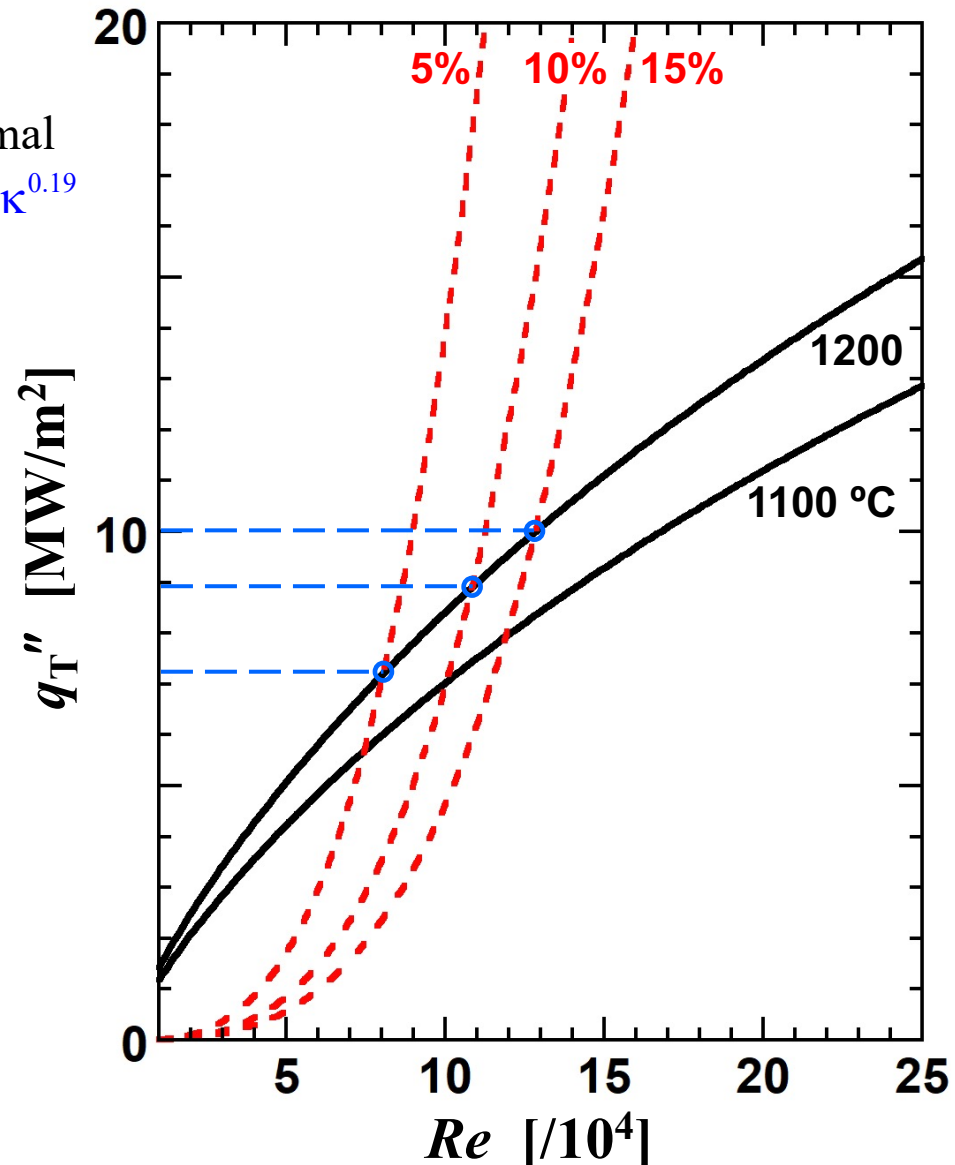
- $K_L \approx 1.84$ , *vs.* 1.73 for He

## ■ Estimate maximum incident heat flux on tile $q_T''$ removed at $T_i = 600$ °C, $p \approx 10$ MPa

- Max. average pressure boundary temperature  $T_s = 1100$  °C, 1200 °C
- sCO<sub>2</sub> pumping power fraction  $\beta = 5\%, 10\%, 15\%$

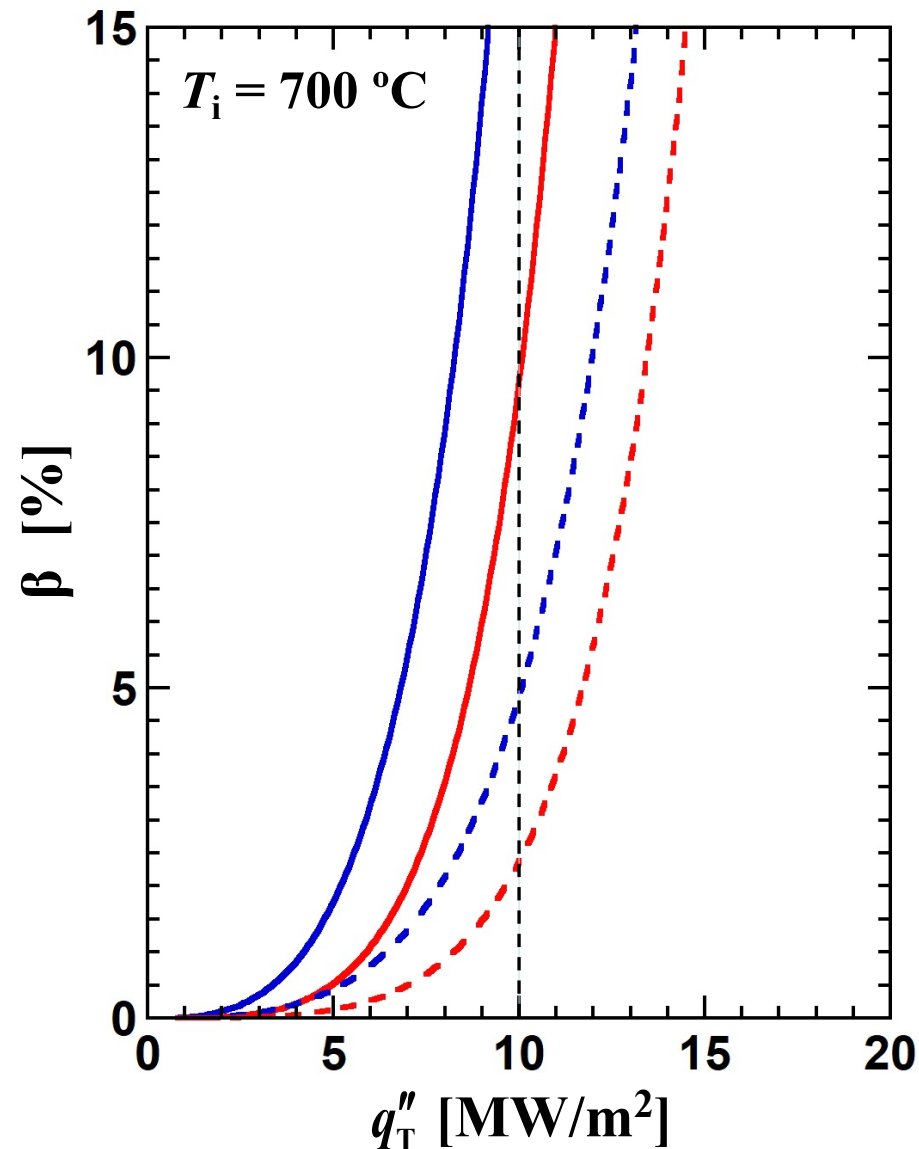
## ■ New performance curves

- For given  $T_s$ , determine  $\beta(q_T'')$ 
  - $Re$  increases as  $\beta$  and  $q_T''$  increase



# NEW PERFORMANCE CURVES

- Pumping power fraction  $\beta$  as a function of max. incident heat flux on tile  $q_T''$  removed for constant  $T_s = 1200\text{ °C}$ 
  - **sCO<sub>2</sub>** vs. **He** for  $p = 10\text{ MPa}$  (—) and  $20\text{ MPa}$  (--) at  $T_s = 1200\text{ °C}$
  - $\beta$  higher for **sCO<sub>2</sub>** for given set of conditions  $\Rightarrow$  **He** has better thermo-fluids performance



# *sCO<sub>2</sub> vs. He*

- Heat flux on tile  $q_T''$  (and  $Re$ ) at  $p_o = 10$  MPa

$T_i = 600$ °C		$\beta = 5\%$		$\beta = 10\%$	
		$T_s = 1100$ °C	$T_s = 1200$ °C	$T_s = 1100$ °C	$T_s = 1200$ °C
	<i>sCO<sub>2</sub></i>	7.0 MW/m <sup>2</sup> (9.9×10 <sup>4</sup> )	8.8 MW/m <sup>2</sup> (1.1×10 <sup>5</sup> )	8.6 MW/m <sup>2</sup> (1.3×10 <sup>5</sup> )	10.8 MW/m <sup>2</sup> (1.4×10 <sup>5</sup> )
	He	9.1 MW/m <sup>2</sup> (2.0×10 <sup>4</sup> )	11.3 MW/m <sup>2</sup> (2.2×10 <sup>4</sup> )	10.5 MW/m <sup>2</sup> (2.7×10 <sup>4</sup> )	13.1 MW/m <sup>2</sup> (2.9×10 <sup>4</sup> )

$T_i = 700$ °C		$\beta = 5\%$		$\beta = 10\%$	
		$T_s = 1100$ °C	$T_s = 1200$ °C	$T_s = 1100$ °C	$T_s = 1200$ °C
	<i>sCO<sub>2</sub></i>	5.2 MW/m <sup>2</sup> (7.8×10 <sup>4</sup> )	6.8 MW/m <sup>2</sup> (8.5×10 <sup>4</sup> )	6.3 MW/m <sup>2</sup> (1.0×10 <sup>5</sup> )	8.3 MW/m <sup>2</sup> (1.1×10 <sup>5</sup> )
	He	6.6 MW/m <sup>2</sup> (1.6×10 <sup>4</sup> )	8.7 MW/m <sup>2</sup> (1.7×10 <sup>4</sup> )	7.8 MW/m <sup>2</sup> (2.1×10 <sup>4</sup> )	10.1 MW/m <sup>2</sup> (2.3×10 <sup>4</sup> )

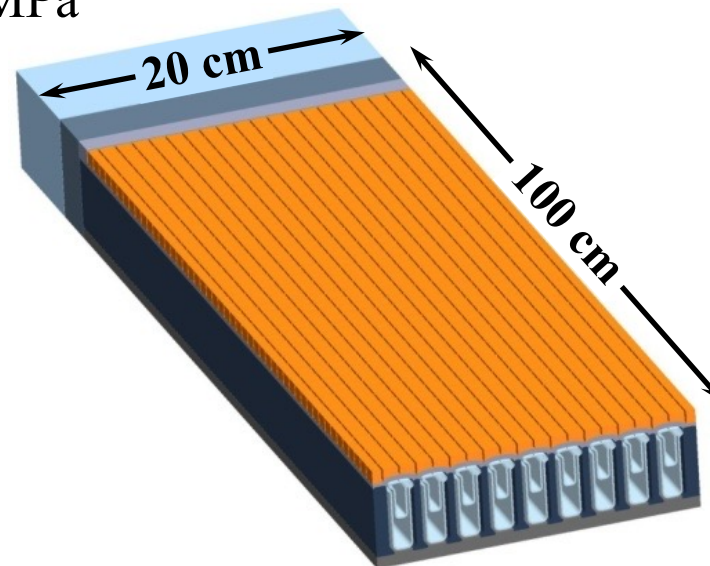
# HE-COOLED FLAT PLATE

## ■ Helium-cooled flat plate (HCFP) divertor

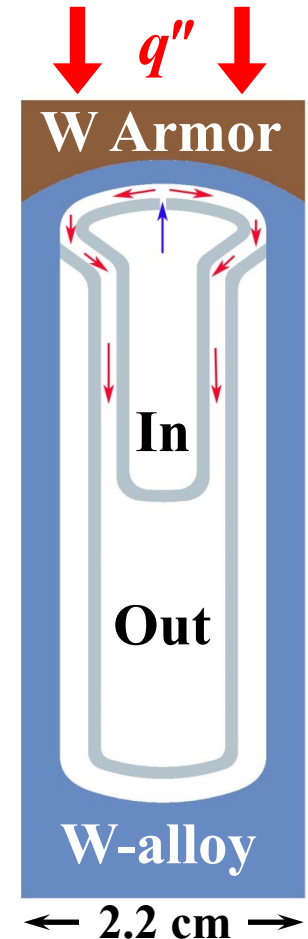
- Cool large areas, but at lower  $q'' < 8 \text{ MW/m}^2$

Hermsmeyer & Malang 02; Wang et al. 09

- $O(10^2)$  plate modules required to cool  $150 \text{ m}^2$  target plate, vs.  $O(10^5)$  “fingers”
- Experimental studies of HCFP in small GT He loop
  - $\dot{m} \leq 10 \text{ g/s} \Rightarrow$  “short” test section with  $L \approx 3 \text{ cm}$  slot to achieve prototypical  $Re = 3.3 \times 10^4$
  - First studies with He at  $p \approx 10 \text{ MPa}$
  - Heat fluxes  $q'' < 2 \text{ MW/m}^2$  from electric cartridge heaters
  - Inlet temperatures  $T_i \leq 100 \text{ }^\circ\text{C}$
  - Experiments complemented by numerical simulations at prototypical conditions



$$Re = \frac{2\dot{m}}{\mu(T_i)L}$$

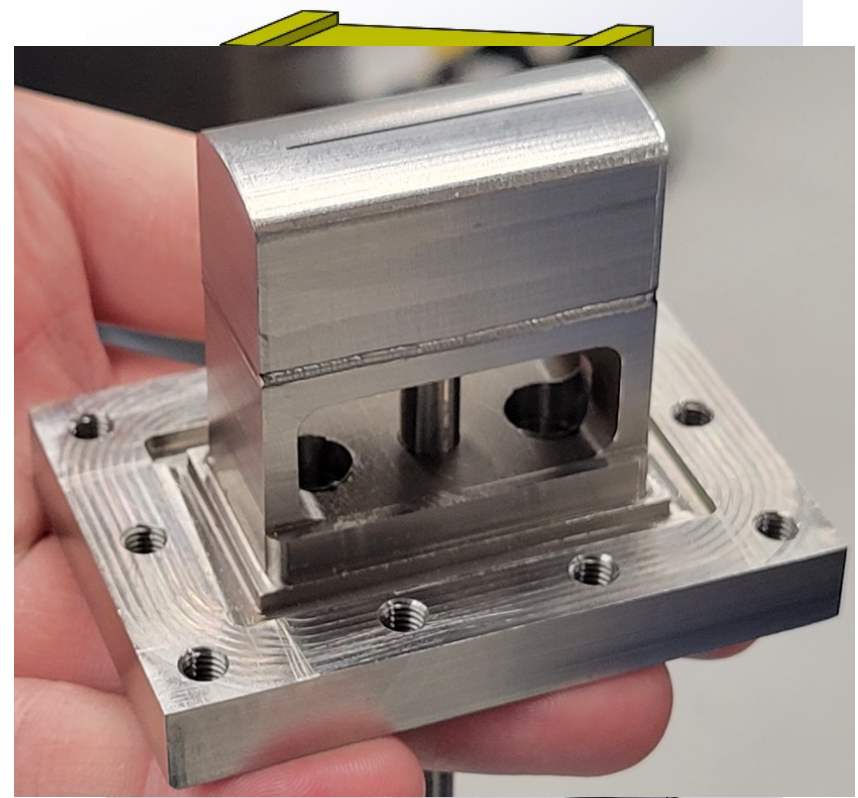
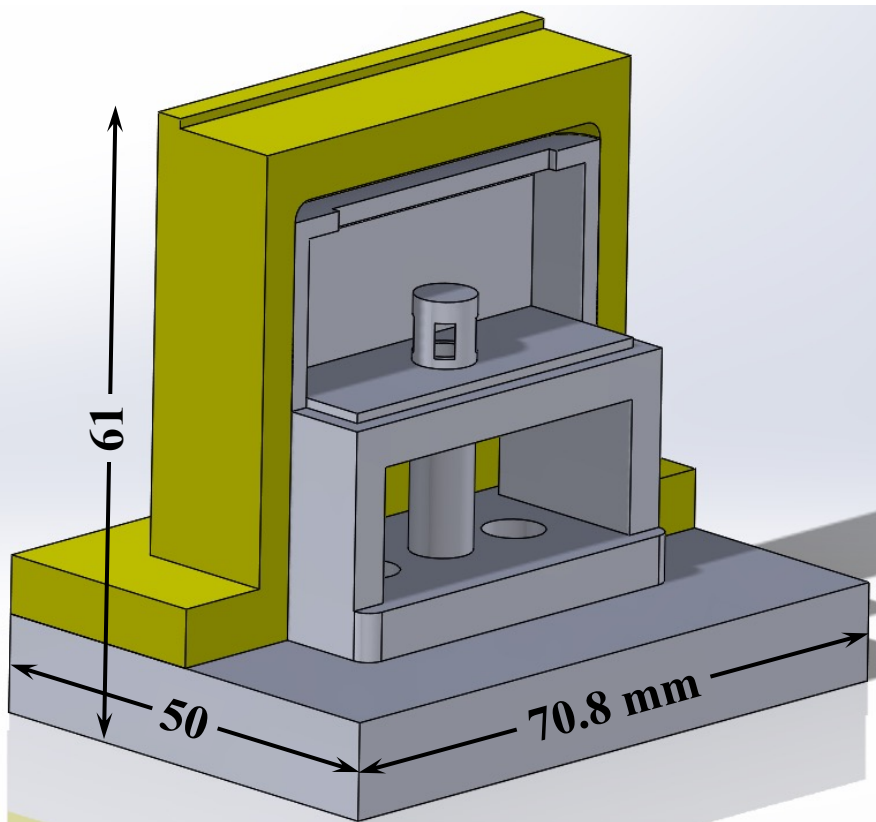




# HCFP TEST SECTION

## ■ Fabricating “short” HCFP test section

- Slot: width  $W = 0.5$  mm (width)  $\times$  length  $L = 29.8$  mm
- Stainless 316 inner cartridge + Ampcoloy 944 (Cu-Ni) outer shell



# ***FUTURE WORK***

## **Next steps**

### ■ **Reexamine performance curves**

- Base on maximum (vs. average) pressure boundary temperature imposed by material limits
- Add limits imposed by maximum thermal stresses

### ■ **Helium-cooled flat plate (HCFP) experiments**

- Develop  $Nu$  and  $K_L$  correlations and performance curves based on experimental data + numerical simulations at prototypical conditions

### ■ **Numerical analysis: integrate multiple divertor units with first wall**

- “Tile” outboard side of first wall sector model with individual T-tube units
- Nonuniform heat flux distribution: curve-fit to simulations by T. D. Rognlien
- Variations in He mass flow rates to each unit